

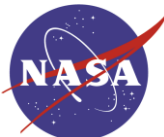
Overview of the Lunar Laser Communications Demonstration

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Don M. Boroson, Andrew S. Fletcher**
MIT Lincoln Laboratory

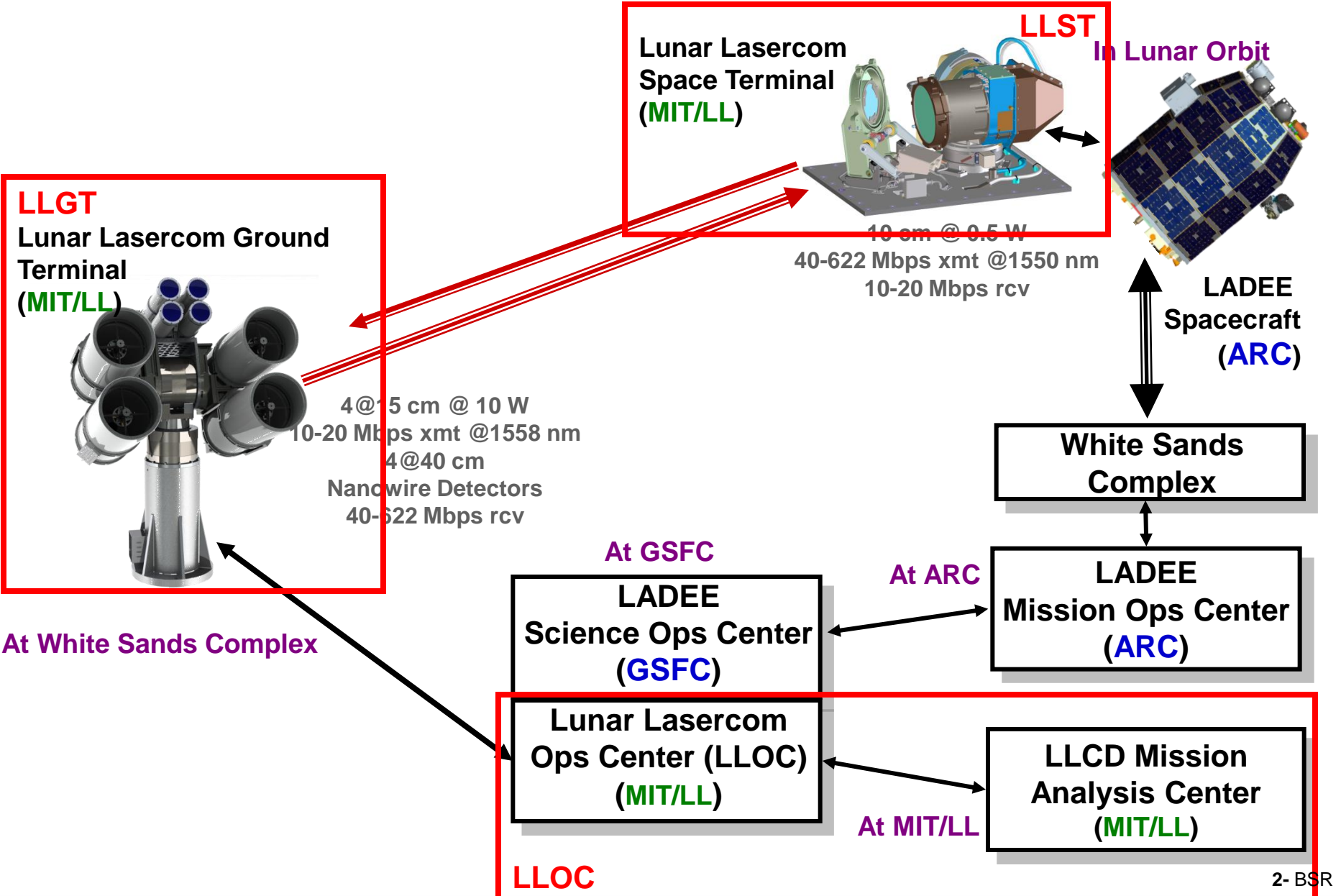
Bernard L. Edwards
NASA Goddard Space Flight Center

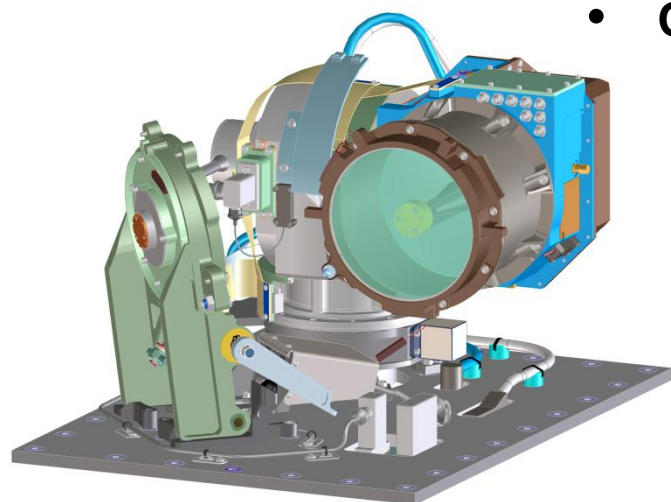
**Optical Coding and Modulation SIG
at CCSDS Fall 2010 meetings
October 25, 2010**

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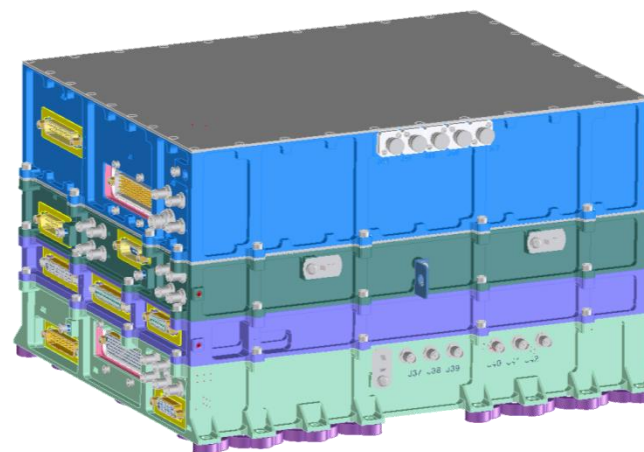


Lunar Laser Communication Demo System



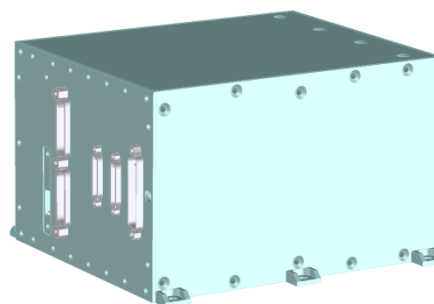


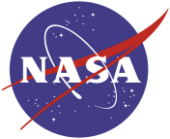
- **Optical Module (OM)**
 - Inertially-stabilized
 - 2-axis gimbal
 - Fiber coupled to modem transmit and receive



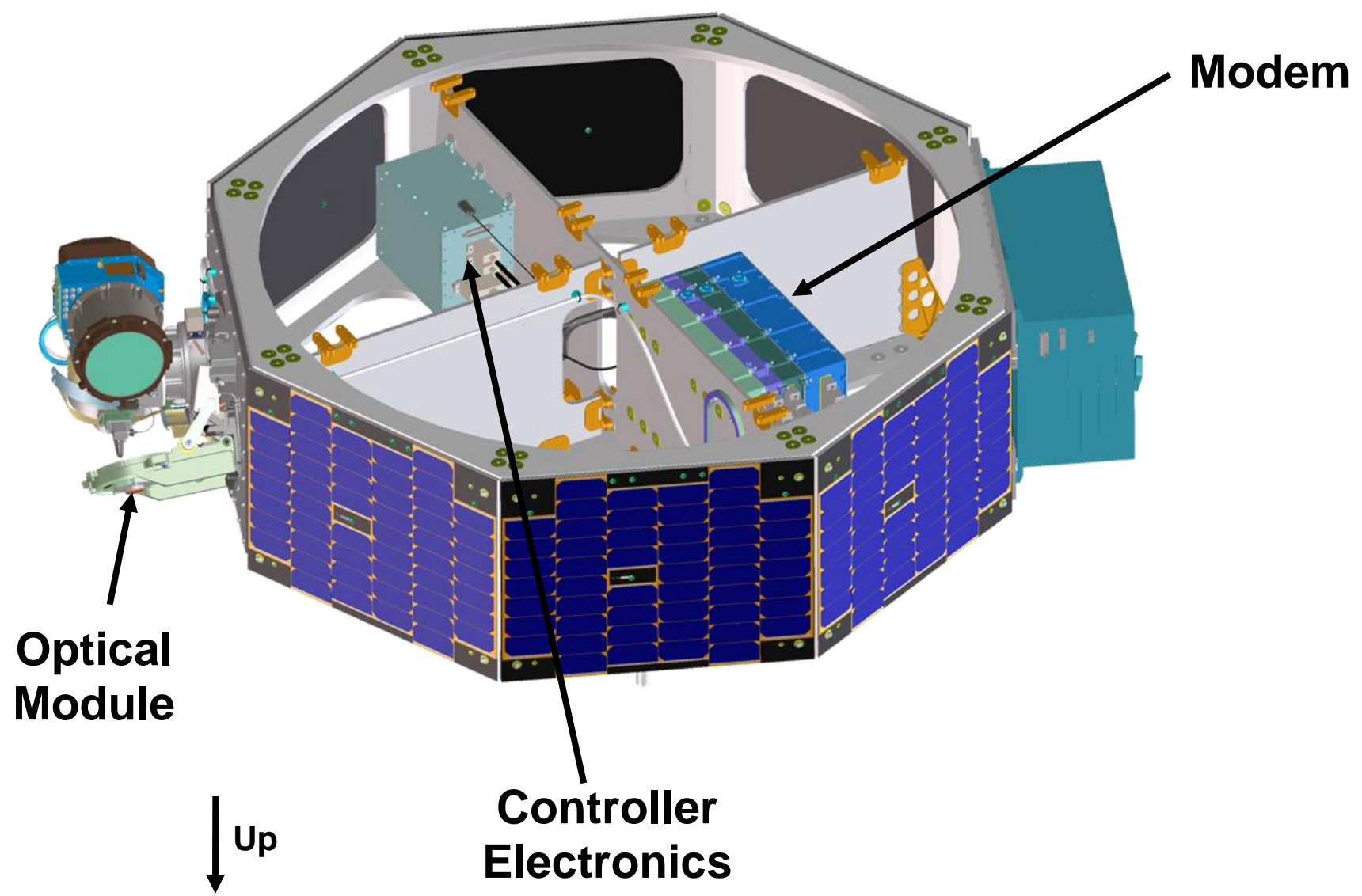
- **Modem Module (MM)**
 - EDFA transmitter
 - Optically-preamplified receiver
 - CODECs and data interfaces
 - Data interface to spacecraft

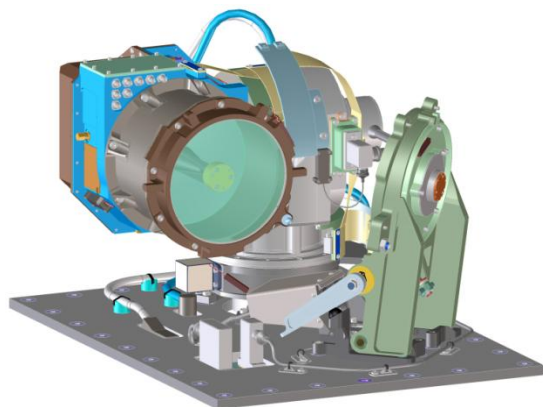
- **Controller Electronics (CE)**
 - OM, MM control
 - CMD/TLM interface to S/C





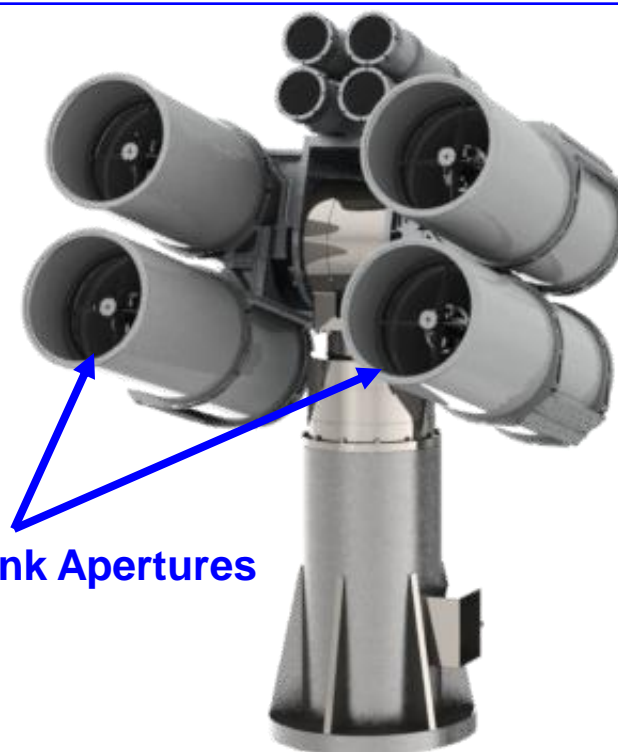
LLST on LADEE





Space Terminal Transmitter

- 10-cm transmit aperture
- Fiber coupled to 0.5-W 1550-nm Master Oscillator Power Amplifier (MOPA) transmitter
- Data rates from 40-620 Mbit/s

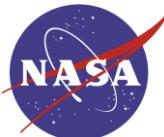


Downlink Apertures

*** FIGURES NOT TO SCALE!**

Ground Terminal Receiver

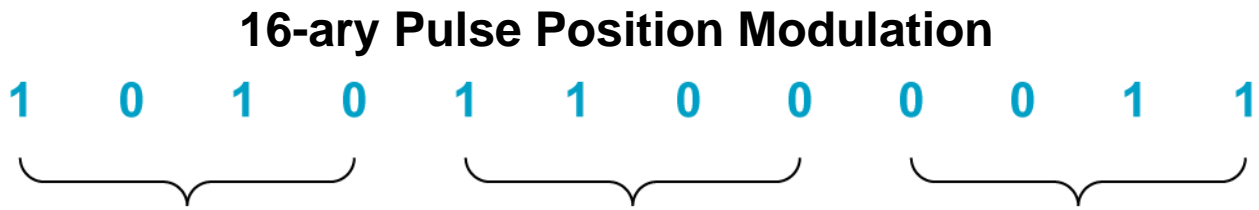
- 4 x 40-cm collection aperture
- Each telescope fiber-coupled to photon-counting detector array



LLCD Downlink Signaling Modulation and Coding

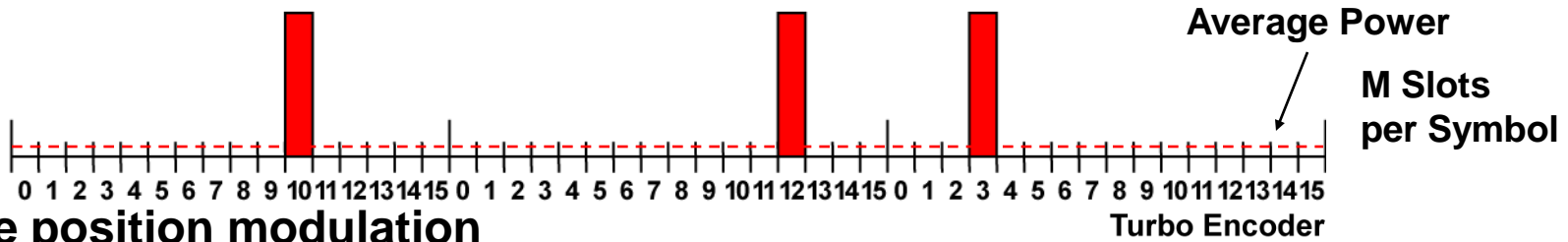


Source Data

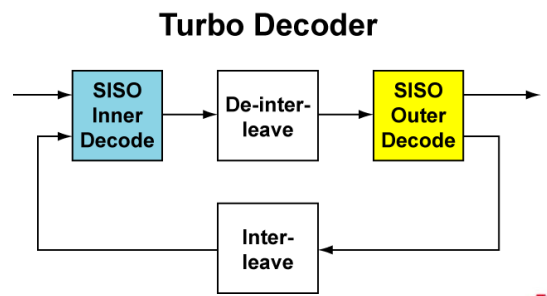
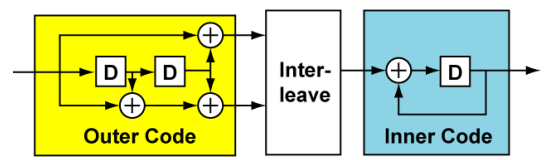


PPM Symbols

PPM Waveform

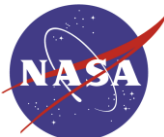


- **Pulse position modulation**
 - Orthogonal, power efficient
 - Well-suited for MOPA transmitter architecture
 - Well-suited to photon-counting detectors
 - Limited count rate (reset time)
 - High timing resolution
- **Data rate varied by changing slot frequency**
 - 311 MHz – 5 GHz
- **1/2-rate serially concatenated turbo code**
 - Simple encoder
 - Iterative decoder
 - Operates within 1 dB of theoretical channel capacity



Serially Concatenated PPM Turbo Code



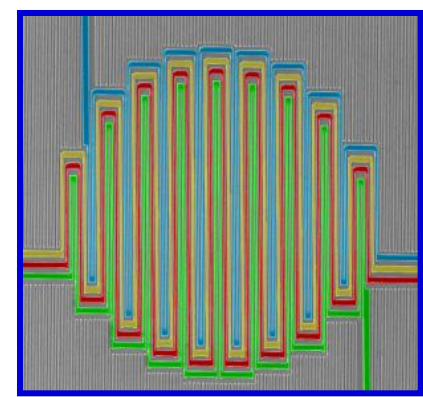
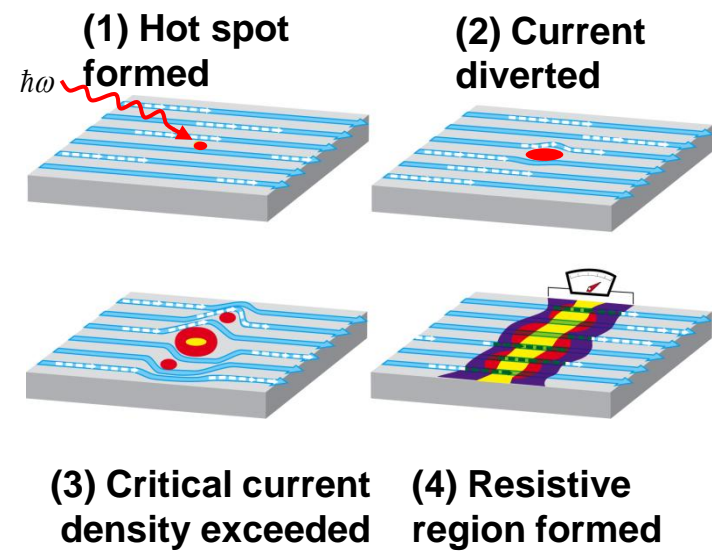


LLCD Superconducting Nanowire Detector Arrays (SNDA)



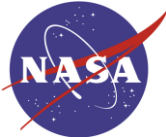
Principle of operation

Nanowire superconducting at $T < 2\text{ K}$

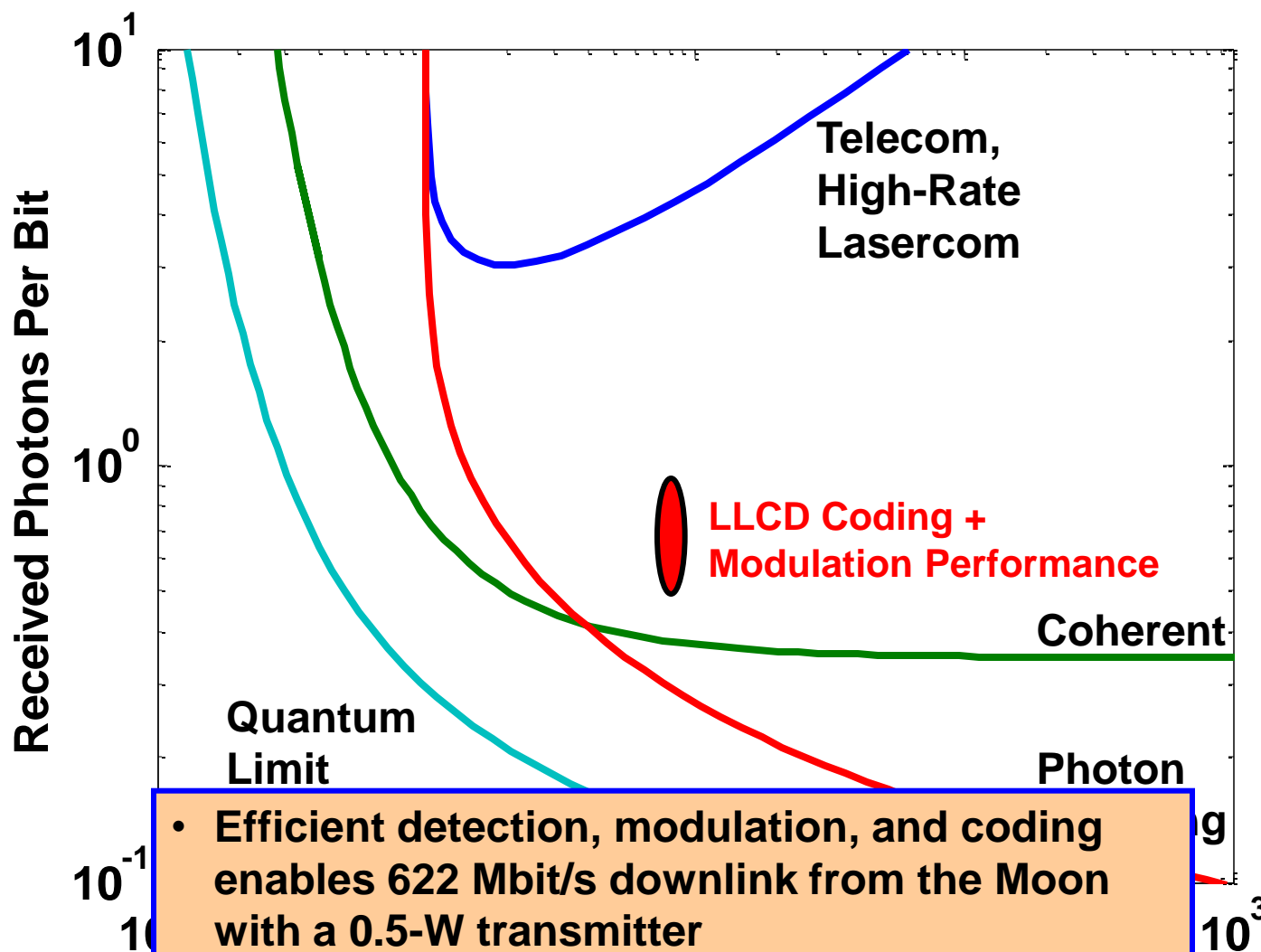


4-element LLCD SNDA

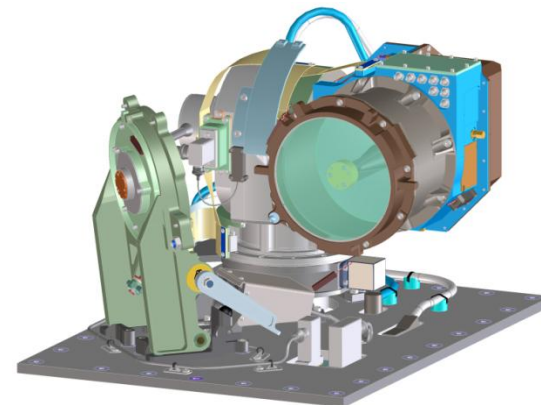
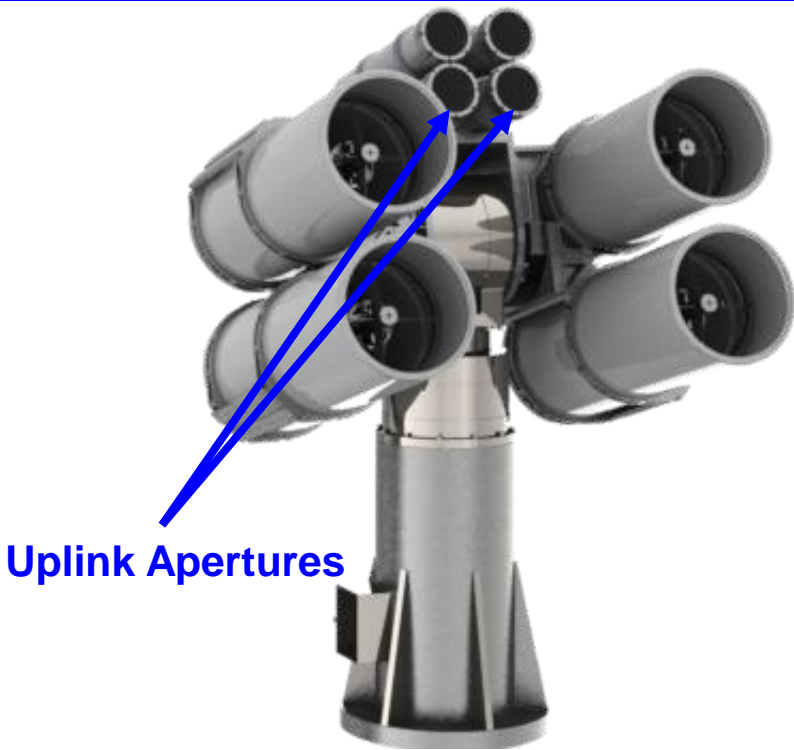
- Superconducting nanowire detects individual photons
 - >60% detection efficiency
 - <15-ns reset time
 - <50-kHz dark count rate
- Detectors coupled to telescopes using custom multi-mode polarization maintaining fiber
- Detector array used to achieve large detection area and high count rates
 - 4 detectors per telescope
 - 4 telescopes



LLCD Downlink Power Efficiency



- Efficient detection, modulation, and coding enables 622 Mbit/s downlink from the Moon with a 0.5-W transmitter
- LLCD transmitter is 4000x more energy efficient (per bit) than LADEE RF transceiver



*** FIGURES NOT TO SCALE!**

Ground Terminal Transmitter

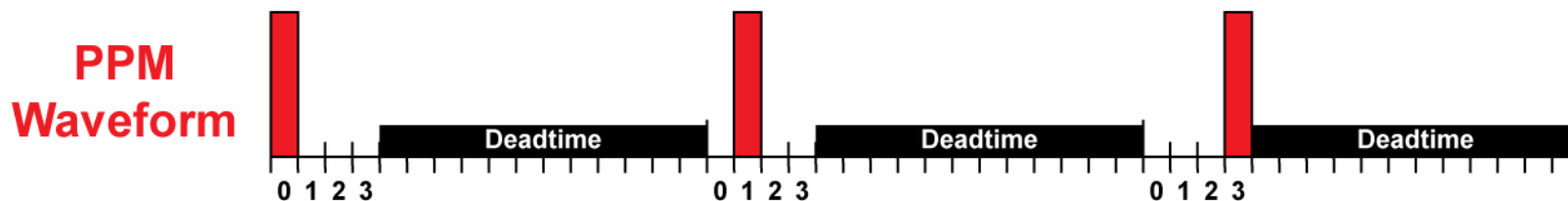
- 4 x 15-cm transmit aperture
- 10-W per transmit aperture (40 W, total)
- Data rates from 10-20 Mbit/s

Space Terminal Receiver

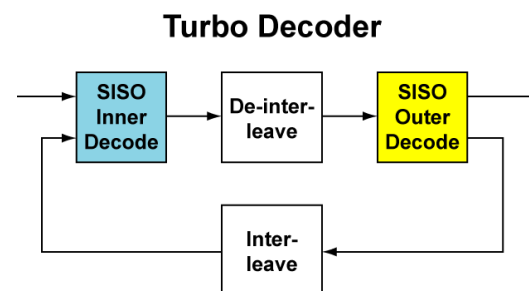
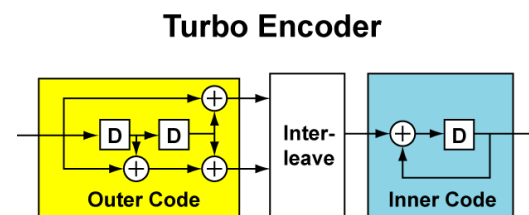
- 10-cm receive aperture
- Non-coherent combining of uplink signals
- Fiber coupled to optically-preamplified direct detection receiver



LLCD Uplink Signaling Modulation, Coding, Detection



- **4-ary pulse position modulation with dead time**
 - 311-MHz slot rate
 - Deadtime provides better match with narrow-band optical filter
 - Data rate varied by changing dead time (10 or 20 Mbit/s)
- **Pre-amplified direct-detection receiver**
- **Near-optimal hard-decision 4-PPM decoder**
- **1/2-rate serially concatenated turbo code**
 - < 1 dB from theoretical channel capacity

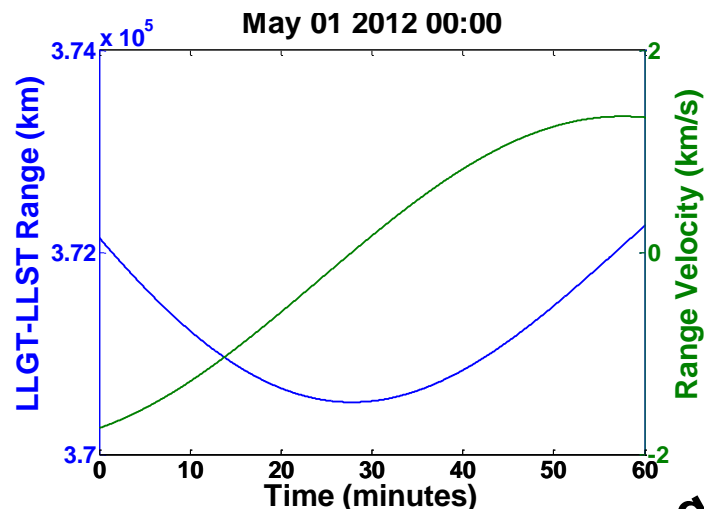


Serially Concatenated PPM Turbo Code

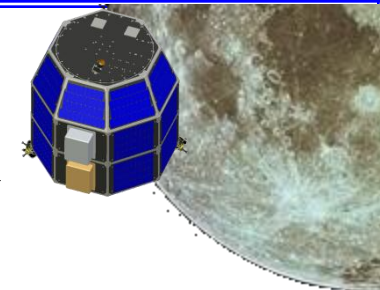




LLGT-LLST Ranging



Range = 350,000 – 400,000 km



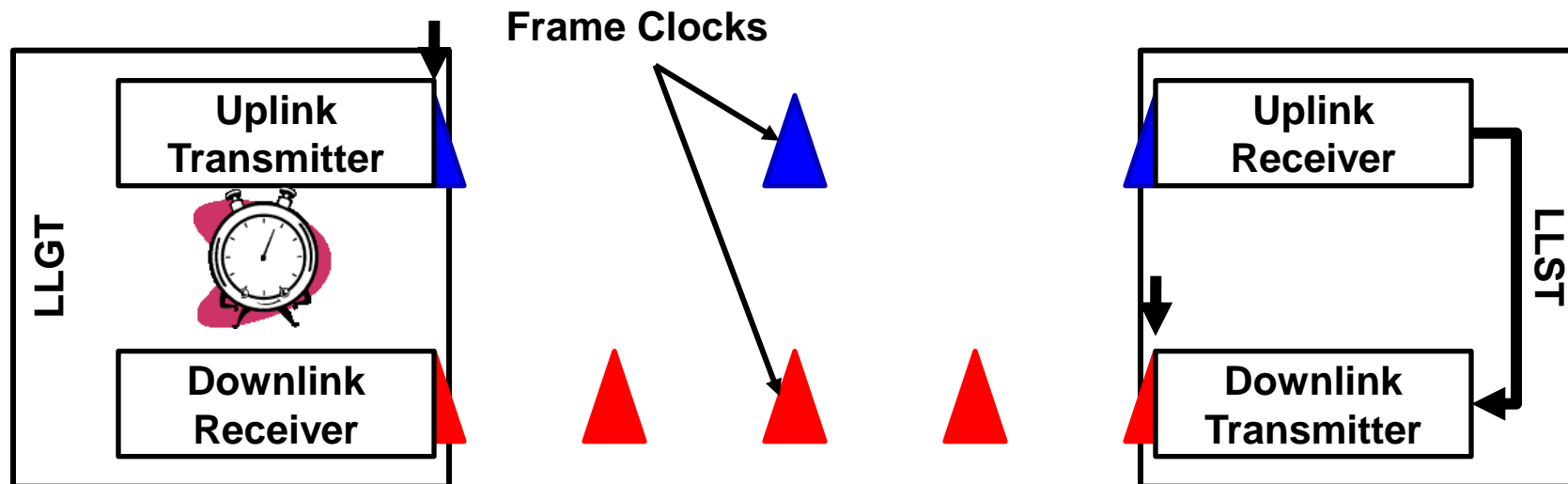
- Ranging is critical for space navigation
- Fast optical clocks on uplink and downlink comm signals potentially enable high-resolution ranging
 - 5-GHz downlink clock → <1-cm range resolution
- To demonstrate this potential, LLCD will measure two-way time-of-flight to sub-slot precision

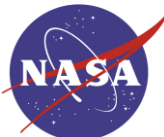


Two-Way Time-of-Flight Measurements



- Duplex communications systems may be easily augmented to perform two-way time-of-flight measurements
- Additional requirements include
 - Common time reference on forward and return links
 - LLCD uplink frame duration = 32 x downlink frame duration
 - Phase-locked clocks in one of the terminals
 - Uplink phase-locked to downlink in LLST
 - High-stability time reference for measuring two-way time-of-flight

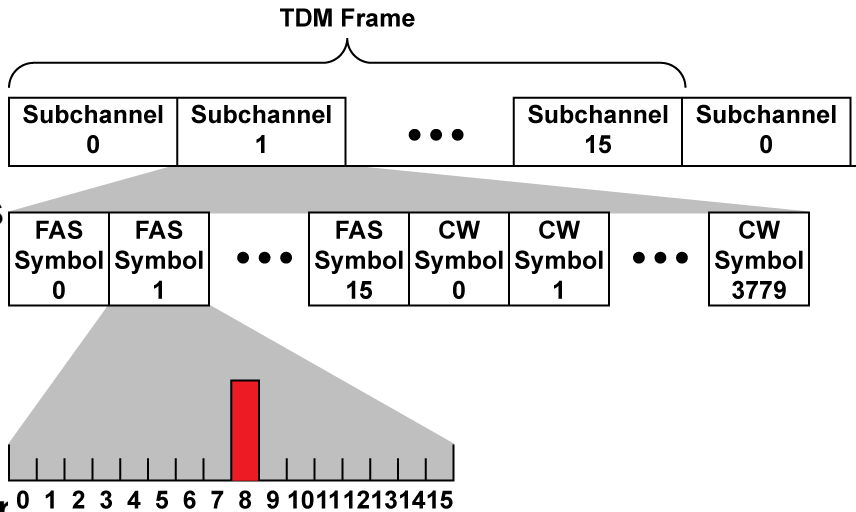




LLCD Clocks for Two-Way Time-of-Flight Measurements

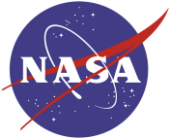


- Numerous clocks embedded in uplink and downlink optical signals
- Successful optical communications requires precise synchronization of clocks
 - Slot
 - Symbol
 - Codeword
 - Frame
- High frequency clocks provide high-resolution common time reference between transmitter and receiver
- Low frequency frame clocks provide unambiguous measurements of long time scales
 - Downlink frame gives 58.5-km ambiguity
 - Can be improved to 1873-km ambiguity by observing loopback frame sequence
 - Could be further improved by inserting information in loopback data stream
- High-stability GPS-disciplined clock at LLGT used to generate uplink clocks and compare with received downlink clocks



	Duration	Distance
Uplink		
Slot	3.2 ns	96 cm
Symbol	51.4 ns	15.4 m
Codeword	390 us	117 km
TDM Frame	6.25 ms	1873 km
Downlink		
Slot	200 ps	6 cm
Symbol	3.2 ns	96 cm
Codeword	12.2 us	3.7 km
		58.5 km

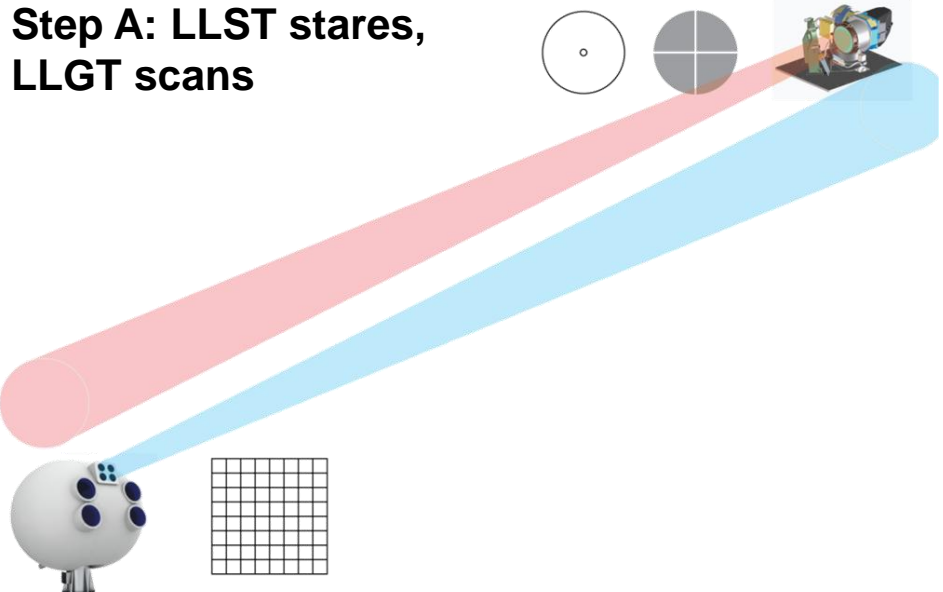
Combination of high-bandwidth modulation and low-frequency frame clocks enables unambiguous sub-cm ranging.



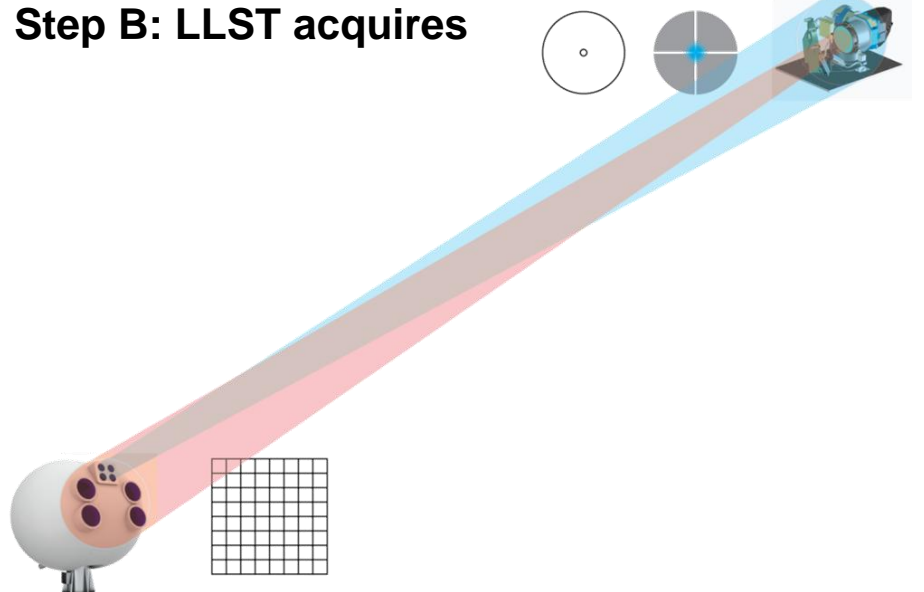
Acquisition Sequence Summary



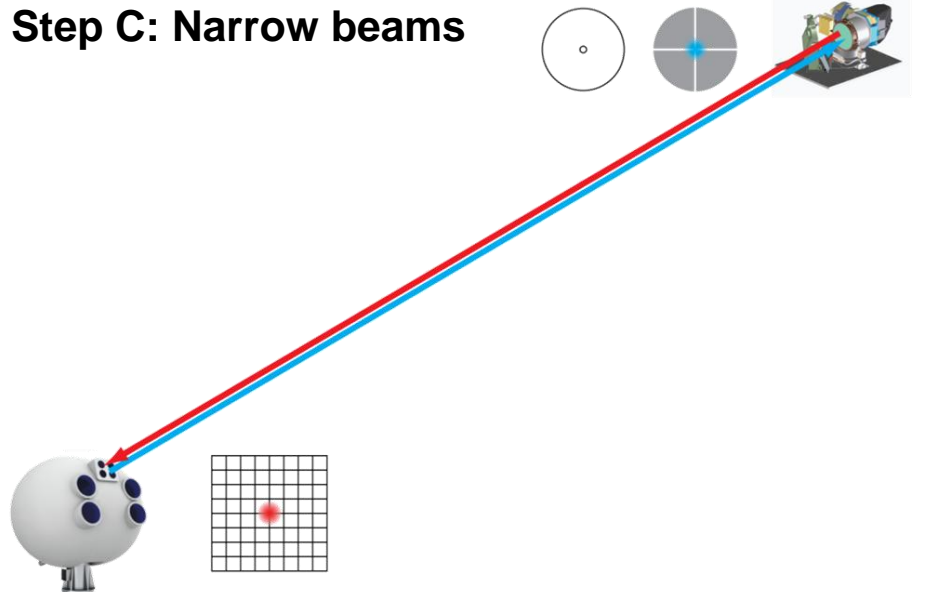
Step A: LLST stares, LLGT scans



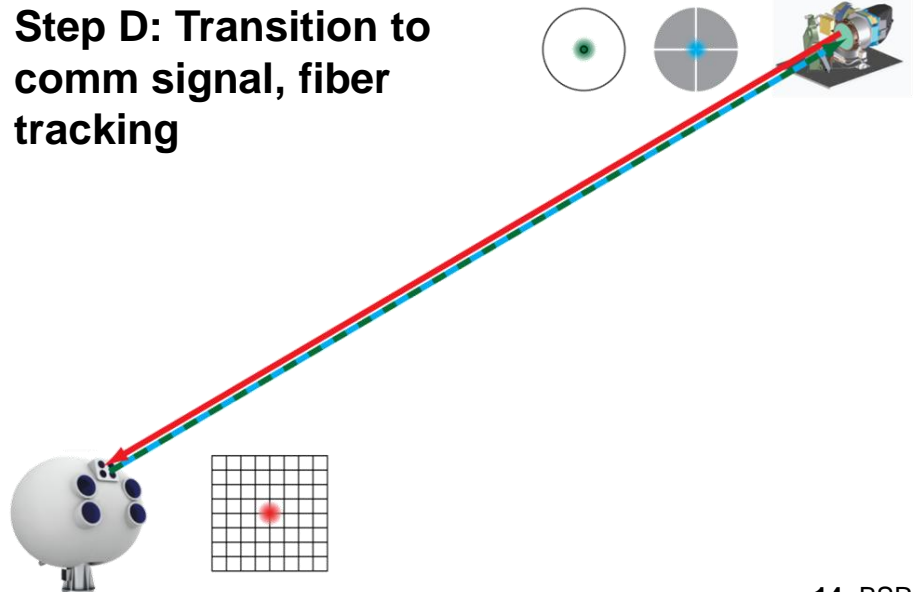
Step B: LLST acquires



Step C: Narrow beams



Step D: Transition to comm signal, fiber tracking





Some Conditions That Can Vary in Lasercom Links



- Ground Terminal elevation – low / medium / high
- Time of day – dark night / dusk-dawn / bright sky

- Ground Terminal SEP – small / other
- Ground Terminal brightness behind Space Terminal – dark / light / terminator
- Space Terminal background – dark / medium / clouds-snow
- Space Terminal SPE – small / other

- Ground Terminal winds – low / medium / high
- Ground Terminal turbulence – low / medium / high
- Ground Terminal visibility – good / medium / poor

- Space Terminal temperature – near 0 / room temp / hot
- Time since last Ground Terminal calibration
- Time since last Space Terminal boresight

- *Hundreds of combinations*
- *LLCD will try to sample as many of these as possible in its short mission*

Summary

- **Optical communications offers the potential for substantial improvements in deep space communications capabilities while reducing burden on spacecraft**
- **LLCD will be NASA's first step in making lasercom a reality for future NASA missions**

